DAFTAR PUSTAKA

- Anderson, J. D. (2010). Fundamental Of Aerodynamics. In *Schweizerische medizinische Wochenschrift*. McGraw hill.
- Andino, M. Y., Lin, J. C., Roman, S., Graff, E. C., Gharib, M., Whalen, E. A., & Wygnanski, I. J. (2019). Active flow control on vertical tail models. *AIAA Journal*, 57(8), 3322–3328. https://doi.org/10.2514/1.J057876
- Angeline Rerung, Z., Sofyan, E., & Setiawan, F. (2020). Analisis Kestabilan Statik Dan Dinamik Pada Pesawat Lsu-05 Ng (Lapan Surveillance Uav 05 New Generation) Dengan Menggunakan Perangkat Lunak Xflr5. *Teknika STTKD: Jurnal Teknik, Elektronik, Engine, 6*(2), 76–83. https://doi.org/10.56521/teknika.v6i2.215
- B. Barlow, J., H. Rae Jr, W., & Pope, A. (1999). Low-speed Wind Tunnel Testing (Third, pp. 301–425).
- Biadgo, A. M., Simonovic, A., Svorcan, J., & Stupar, S. (2014).
 Aerodynamic characteristics of high speed train under turbulent cross Winds: A numerical investigation using unsteady-RANS method. *FME Transactions*, 42(1), 10–18. https://doi.org/10.5937/fmet1401010B
- Cao, X., Dong, H., Gu, Y., Cheng, K., & Zhang, F. (2023). Experimental Study of Vertical Tail Model Flow Control Based on Oscillating Jet. *Applied Sciences (Switzerland)*, 13(2). https://doi.org/10.3390/app13020786
- Cengel, Y. A., & Cimbala, J. M. (2014). *Fluid mechanics*: *Fundamental and Aplication*. McGraw-Hill.
- Ciliberti, D., Della Vecchia, P., Nicolosi, F., & De Marco, A. (2017). Aircraft directional stability and vertical tail design: A review of semi-empirical methods. *Progress in Aerospace Sciences*, 95(November), 140–172.

https://doi.org/10.1016/j.paerosci.2017.11.001

- Corcione, S., Cusati, V., Memmolo, V., Nicolosi, F., & Llamas Sandin, R. (2023). Impact at aircraft level of elastic efficiency of a forwardswept tailplane. *Aerospace Science and Technology*, 140, 108461. https://doi.org/10.1016/j.ast.2023.108461
- Corte, B. Della, Sluis, M. Van, Rao, A. G., & Veldhuis, L. (2019). Experimental Investigation of the Flow Past an Axisymmetric Body at Low Speed. *Proceedings of the International Society of Air*

Breathing Engines, *September*, 1–22. https://www.semanticscholar.org/paper/Experimental-Investigation-of-the-Flow-Past-an-Body-Corte-Sluis/a696c906c47580b696e8e2f48f123ced69c03de4

- Das, S. B., Rajiv, R., Menon, R., & Deodhar, R. (2021). Analysis and simulation of different empenage configurations for an aircraft. *AIP Conference Proceedings*, 2316(February). https://doi.org/10.1063/5.0038266
- Elger, D. F., Williams, B. C., Crowe, C. T., & Roberson, J. A. (2012). Engineering fluid mechanics (10th ed.). WileyPLUS.
- Erturk, S. A., & Dogan, A. (2017). Trim analyses of mass-actuated airplane in cruise and steady-state turn. *Journal of Aircraft*, 54(4), 1585–1592. https://doi.org/10.2514/1.C034200
- Fu, J., Shi, Z., Gong, Z., Lowenberg, M. H., Wu, D., & Pan, L. (2022). Virtual flight test technique to predict a blanded wing-body aircraft in-flight depature characteristic. *Chinese Journal of Aeronautics*, 35(1), 215–225. https://doi.org/10.1016/j.cja.2021.01.006
- FU, J., SHI, Z., GONG, Z., LOWENBERG, M. H., WU, D., & PAN, L. (2022). Virtual flight test techniques to predict a blended-wing-body aircraft in-flight departure characteristics. *Chinese Journal of Aeronautics*, 35(1), 215–225. https://doi.org/10.1016/j.cja.2021.01.006
- Grauer, J. A., & Boucher, M. J. (2019). Identification of aeroelastic models for the X-56A longitudinal dynamics using multisine inputs and output error in the frequency domain. *Aerospace*, 6(2). https://doi.org/10.3390/aerospace6020024
- Guo, L., Zhu, M., Nie, B., Kong, P., & Zhong, C. (2017). Initial virtual flight test for a dynamically similar aircraft model with control augmentation system. *Chinese Journal of Aeronautics*, 30(2), 602– 610. https://doi.org/10.1016/j.cja.2016.12.034
- Hartwich, P. M., Camacho, P. P., El-Gohary, K., Gonzales, A. B., Lawson, E. L., & Shmilovich, A. (2017). System-level trade studies for transonic transports with active flow control (AFC) enhanced high-lift systems. AIAA SciTech Forum - 55th AIAA Aerospace Sciences Meeting. https://doi.org/10.2514/6.2017-0321
- Houghton, E. L., Carpenter, P. W., Collicott, S. H., & Valentine, D. (2012). Aerodynamics for Engineering Students. Elsevier. http://journal.um-surabaya.ac.id/index.php/JKM/article/view/2203
- Hu, Y., Song, J., & Wu, M. (2024). A Review of Control Methods for Tailless Aircraft. *Guidance, Navigation and Control, 2430002.*

https://doi.org/10.1142/S2737480724300026

- Ignatyev, D. I., Zaripov, K. G., Sidoryuk, M. E., Kolinko, K. A., & Khrabrov, A. N. (2016). Wind tunnel tests for validation of control algorithms at high angles of attack using autonomous aircraft model mounted in 3DOF gimbals. *AIAA Atmospheric Flight Mechanics Conference*, 2016-Janua(June), 1–18. https://doi.org/10.2514/6.2016-3106
- Jansen, K. E., Rasquin, M., Farnsworth, J. A., Rathay, N., Monastero, M. C., & Amitay, M. (2018). Interaction of a synthetic jet with separated flow over a vertical tail. *AIAA Journal*, 56(7), 2653– 2668. https://doi.org/10.2514/1.J056751
- Khan, M. U., Khan, M. D., Din, N. A., Babar, M. Z., & Hussain, M. F. (2019). Aerodynamic comparison of unconventional aircraft tail setup. *Proceedings 22nd International Multitopic Conference*, *INMIC* 2019, 1–5. https://doi.org/10.1109/INMIC48123.2019.9022788
- Kreith, F. (2012). *Principles Of Heat Transfer* (Third Edit). New York and London.
- Kreith, F., & Guswami, D. Y. (2005). The CRC Handbook Of Engineering Mechanica, Second Edition. In Interior Finishes & Fittings for Historic Building Conservation (Second). CRC Press. https://doi.org/10.1002/9781444344837.ch10
- Leelaburanathanakul, P., Virangkur, V., Wangsiripaisarn, T., Pitakarnnop, J., & Bunyajitradulya, A. (2021). Steady tangential control jet for improving the effectiveness of a rudder under oneengine inoperative condition. *Journal of Physics: Conference Series*, 1733(1). https://doi.org/10.1088/1742-6596/1733/1/012001
- Liu, Y., & Xie, C. (2018). Aeroservoelastic stability analysis for flexible aircraft based on a nonlinear coupled dynamic model. *Chinese Journal of Aeronautics*, 31(12), 2185–2198. https://doi.org/10.1016/j.cja.2018.08.019
- Liu, Z., Luo, L., & Zhang, B. (2021). An aerodynamic design method to improve the high-speed performance of a low-aspect-ratio tailless aircraft. *Applied Sciences (Switzerland)*, 11(4), 1–25. https://doi.org/10.3390/app11041555
- Löffler, S., Staats, M., Grund, T., & Weiss, J. (2018). Increasing the effectiveness of a vertical stabilizer by combining pulsed jet actuation at the leading edge and the rudder hinge line. 2018 Applied Aerodynamics Conference. https://doi.org/10.2514/6.2018-2854

- Merryisha, S., & Rajendran, P. (2019). Experimental and cfd analysis of surface modifiers on aircraft wing: A review. *CFD Letters*, *11*(10), 46–56.
- Mi, B. gang. (2021). Simulation on the dynamic stability derivatives of battle-structure-damaged aircrafts. *Defence Technology*, 17(3), 987–1001. https://doi.org/10.1016/j.dt.2020.06.005
- Mokhtari, A., Shahrian, A., Langroodi, P. J., & Ghodrat, M. (2020). Investigation of the effects of angle of attack and tail deflection angle on the controlling tail flow field. *Aerospace Systems*, *3*(4), 309–326. https://doi.org/10.1007/s42401-020-00063-w
- Morelli, E. A. (2012). Flight test maneuvers for efficient aerodynamic modeling. *Journal of Aircraft*, 49(6), 1857–1867. https://doi.org/10.2514/1.C031699
- Muchammad. (2019). Analisis momen poros dan gaya samping horn rudder bidang kendali pesawat N-XXX menggunakan computational fluid dynamic. *Paper Knowledge*. *Toward a Media History of Documents*, 15(1), 64–69.
- Munson, B. R., Young, D. F., Okiishi, T. H., & Huebsch, W. W. (2009). Fundamental Of Fluid Mechanics. In *John Wiley & Sons, Inc.* http://civilcafe.weebly.com/uploads/2/8/9/8/28985467/fluid_mech anics.pdf
- Nelson, R. C. (1998). *Flight Stability and Automatic Control*. http://home.eng.iastate.edu/~shermanp/AERE355/lectures/Flight_ Stability_and_Automatic_Control_N.pdf
- Nguyen Van, E., Alazard, D., Döll, C., & Pastor, P. (2021). Co-design of aircraft vertical tail and control laws with distributed electric propulsion and flight envelop constraints. *CEAS Aeronautical Journal*, *12*(1), 101–113. https://doi.org/10.1007/s13272-020-00481-8
- Nicolosi, F., Ciliberti, D., Della Vecchia, P., & Corcione, S. (2020). Experimental analysis of aircraft directional control effectiveness. *Aerospace Science and Technology*, 106(July). https://doi.org/10.1016/j.ast.2020.106099
- Obert, E. (2009). Aerodynamic Design of Transport Aircraft. Aerodynamic Design of Transport Aircraft. https://books.google.co.id/books?id=V1DuJfPov48C&lpg=PP1& hl=id&pg=PR4#v=onepage&q&f=false
- Oleinik, O. A., & Samokhin, V. N. (1999). *Mathematical Models in Boundary Layer Theory* (1st ed., Vol. 15). CRC Press. http://journal.um-surabaya.ac.id/index.php/JKM/article/view/2203

- Parancheerivilakkathil, M. S., Pilakkadan, J. S., Ajaj, R. M., Amoozgar, M., Asadi, D., Zweiri, Y., & Friswell, M. I. (2024). A review of control strategies used for morphing aircraft application. *Chinese Journal of Aeronautics*, 37(4), 436–463. https://doi.org/10.1016/j.cja.2023.12.035
- Rabeta, B., Arifin, M., & Fairuza, S. (2017). Analisis Linear Statik Pada Vertical Tail dengan Variasi Defleksi Rudder. *Jtk: Jurnal Teknologi Kedirgantaraan*, 2(2), 8–16.
- Ricco, P., Skote, M., & Leschziner, M. A. (2021). A review of turbulent skin-friction drag reduction by near-wall transverse forcing. *Progress in Aerospace Sciences*, 123. https://doi.org/10.1016/j.paerosci.2021.100713
- Sanchez-Carmona, A., & Cuerno-Rejado, C. (2019). Vee-tail conceptual design criteria for commercial transport aeroplanes. *Chinese Journal of Aeronautics*, 32(3), 595–610. https://doi.org/10.1016/j.cja.2018.06.012
- Schlichting, H., & Gersten, K. (2017). Boundary-Layer Theory. In *Laboratory Animal Science* (9th editio, Vol. 42, Issue 3). Springer.
- Scholz, P., Singh, V. M., Gebhardt, A., Kirz, J., Löffler, S., & Weiss, J. (2020). The efficiency of different flow control methods on a vertical tail. AIAA Scitech Forum, 1–16. https://doi.org/10.2514/6.2020-1537
- Shi, Z. (2024). Improved FPA for aircraft conceptual design. *Journal* of Engineering Research (Kuwait), November 2023. https://doi.org/10.1016/j.jer.2024.05.002
- Soler, M. (2014). *Fundamentals Of Aerospace Engineering*. Creative Commons.
- Sonwane, P., Yadav, M., Bunker, N., Sonwane, S., & Shirsath, V. (2024). Economical Design Perspective for Aircraft by Optimizing Airfoil S1223. *E-Prime - Advances in Electrical Engineering, Electronics and Energy*, 8(March), 100531. https://doi.org/10.1016/j.prime.2024.100531
- Tai, S., Bu, C., Wang, Y., Yue, T., Liu, H., & Wang, L. (2024). Identification of aircraft longitudinal aerodynamic parameters using an online corrective test for wind tunnel virtual flight. *Chinese Journal of Aeronautics*. https://doi.org/10.1038/srep32868
- Tai, S., Wang, L., Wang, Y., Lu, S., Bu, C., & Yue, T. (2023). Identification of Lateral-Directional Aerodynamic Parameters for Aircraft Based on a Wind Tunnel Virtual Flight Test. Aerospace, 10(4). https://doi.org/10.3390/aerospace10040350

- Wang, L., Tai, S., Yue, T., Liu, H., Wang, Y., & Bu, C. (2022). Longitudinal Aerodynamic Parameter Identification for Blended-Wing-Body Aircraft Based on a Wind Tunnel Virtual Flight Test. *Aerospace*, 9(11). https://doi.org/10.3390/aerospace9110689
- Wang, L., Zhang, N., Liu, H., & Yue, T. (2022). Stability characteristics and airworthiness requirements of blended wing body aircraft with podded engines. In *Chinese Journal of Aeronautics* (Vol. 35, Issue 6, pp. 77–86). https://doi.org/10.1016/j.cja.2021.09.002
- Wei, Z., Li, J., Tang, S., & Yang, Z. (2022). Investigation and Improvement of T-Tail Junction Flow Separation for a Demonstration Aircraft. Aerospace, 9(10). https://doi.org/10.3390/aerospace9100567
- Welty, J. R., Wilson, R. E., Wicks, C. E., & Rorrer, G. L. (2008). Fundamentals of Momentum, Heat, and Mass Transfer. Wiley & Sons.
- Whalen, E. A., Shmilovich, A., Spoor, M., Tran, J., Vijgen, P., Lin, J. C., & Andino, M. (2018). Flight test of an active flow control enhanced vertical tail. *AIAA Journal*, 56(9), 3393–3398. https://doi.org/10.2514/1.J056959
- White, F. M. (2002). Fluid Mechanics (Fourth Edi). McGraw-Hill.
- Wu, Z., Cao, Y., & Ismail, M. (2019). Gust loads on aircraft. *Aeronautical Journal*, 123(1266), 1216–1274. https://doi.org/10.1017/aer.2019.48
- Xi, X., Liu, Y., Xue, P., Liu, X., Bai, C., Zhang, X., & Gao, L. (2024). High-speed multi-camera videogrammetric measurement of fullfield 3D motion and deformation in full-scale crash testing of typical civil aircraft. *Aerospace Science and Technology*, 109375. https://doi.org/10.1016/j.ast.2024.109375
- Zhao, Z., Luo, Z., Deng, X., Zhang, J., Dong, Z., Liu, J., & Li, S. (2023). Novel yaw effector of a flying wing aircraft based on reverse dual synthetic jets. *Chinese Journal of Aeronautics*, 36(12), 151–163. https://doi.org/10.1016/j.cja.2023.06.023

LAMPIRAN

Lampiran A

Tabel A. 1 Perrubahan kecepatan upstream

| No | U (m/s) |
|----|---------|
| 1 | 10 |
| 2 | 12 |
| 3 | 14 |
| 4 | 16 |
| 5 | 18 |
| 6 | 20 |
| 7 | 22 |

Tabel A. 2 Perubahan sudut defleksi rudder

| No | Sudut defleksi <i>rudder</i> (δ) |
|----|----------------------------------|
| 1 | 0° |
| 2 | 5° |
| 3 | 10° |
| 4 | 15° |
| 5 | 20° |
| 6 | 25° |
| 7 | 30° |
| 8 | 35° |
| 9 | 40° |
| 10 | 45° |

Tabel A. 3 Luas proyeksi stabilizer dan rudder untuk tiap perubahan sudut

| Sudut defleksi <i>rudder</i> (δ) A (m^2) |
|---|
| |
| 0 ° 0.000806 |
| 5° 0.001189 |
| 10° 0.001573 |
| 15° 0.001953 |
| 20° 0.002325 |
| 25° 0.002686 |
| 30° 0.003034 |
| 35° 0.003365 |
| 40° 0.003676 |
| 45° 0.003967 |

| δ | b (m) |
|-------------|-----------------------|
| 00 | 0,006 |
| 5° | 0,010 |
| 10° | 0,013 |
| 15° | 0,017 |
| 20° | 0,020 |
| 25° | 0,024 |
| 30 ° | 0,027 |
| 35° | 0,030 |
| 40° | 0,033 |
| 45° | 0,035 |

Tabel A. 4 Rentang karakteristik vertikal stabilizer dan *rudder* untuk tiap per<u>ubahan sudut defleksi</u>

Tabel A. 5 Kecepatan *upstream* (U) dan bilangan Reynolds (Re) untuk tiap perubahan sudut defleksi *rudder*

| δ | | L | | U (m/s) | | | |
|-------------|-------|----------|-------|------------|-------|-------|-------|
| - | 10 | 12 | 14 | 16 | 18 | 20 | 22 |
| - | | | Bila | angan Reyn | olds | | |
| 0 ° | 3687 | 4425 | 5162 | 5900 | 6637 | 7375 | 8112 |
| 5° | 5956 | 7147 | 8338 | 9529 | 10720 | 11912 | 13103 |
| 10 ° | 8193 | 9832 | 11471 | 13109 | 14748 | 16387 | 18025 |
| 15° | 10382 | 12459 | 14535 | 16612 | 18688 | 20765 | 22841 |
| 20° | 12506 | 15008 | 17509 | 20010 | 22512 | 25013 | 27514 |
| 25° | 14634 | 17561 | 20487 | 23414 | 26341 | 29268 | 32194 |
| 30° | 16591 | 19910 | 23228 | 26546 | 29865 | 33183 | 36501 |
| 35° | 18330 | 21996 | 25662 | 29328 | 32994 | 36660 | 40326 |
| 40 ° | 20039 | 24047 | 28055 | 32063 | 36071 | 40078 | 44086 |
| 45° | 21610 | 25932 | 30254 | 34576 | 38898 | 43220 | 47542 |

| U | (δ) | | | | | | | | | |
|-------|---------|-------|-------------|-------|-------|-------|-------------|-------|-------------|-------|
| (m/s) | | | | | | | | | | |
| | 0° | 5° | 10 ° | 15° | 20° | 25° | 30 ° | 35° | 40 ° | 45° |
| 10 | 0.030 | 0.048 | 0.070 | 0.095 | 0.120 | 0.150 | 0.180 | 0.204 | 0.230 | 0.250 |
| 12 | 0.040 | 0.064 | 0.093 | 0.127 | 0.163 | 0.205 | 0.247 | 0.283 | 0.320 | 0.350 |
| 14 | 0.050 | 0.080 | 0.120 | 0.162 | 0.210 | 0.270 | 0.320 | 0.370 | 0.420 | 0.460 |
| 16 | 0.062 | 0.100 | 0.150 | 0.200 | 0.260 | 0.340 | 0.400 | 0.460 | 0.520 | 0.580 |
| 18 | 0.075 | 0.120 | 0.180 | 0.240 | 0.310 | 0.410 | 0.490 | 0.560 | 0.640 | 0.720 |
| 20 | 0.087 | 0.140 | 0.210 | 0.280 | 0.360 | 0.480 | 0.570 | 0.660 | 0.770 | 0.860 |
| 22 | 0.100 | 0.160 | 0.240 | 0.330 | 0.420 | 0.550 | 0.660 | 0.780 | 0.900 | 1.010 |

Tabel A. 6 Nilai gaya drag (F_D) pendekatan eksperimental untuk tiap tingkat perubahan sudut defleksi *rudder* pada 7 tingkat kecepatan

Tabel A. 7 Nilai gaya *side* (F_S) pendekatan eksperimental untuk tiap tingkat perubahan sudut defleksi *rudder* pada 7 tingkat kecepatan

| $\begin{array}{c ccccccccccccccccccccccccccccccccccc$ | Fs | | | | | | | | |
|--|----|--|--|--|--|--|--|--|--|
| 0° 5° 10° 15° 20° 25° 30° 35° 40° 44° 10 0.000 0.030 0.080 0.130 0.196 0.240 0.260 0.300 0.320 0.3 | | | | | | | | | |
| 10 0.000 0.030 0.080 0.130 0.196 0.240 0.260 0.300 0.320 0.3 | 50 | | | | | | | | |
| | 40 | | | | | | | | |
| 12 0.000 0.050 0.120 0.200 0.290 0.360 0.390 0.440 0.480 0.5 | 00 | | | | | | | | |
| 14 0.000 0.080 0.170 0.280 0.400 0.500 0.540 0.620 0.670 0.7 | 10 | | | | | | | | |
| 16 0.000 0.110 0.230 0.380 0.550 0.680 0.740 0.840 0.900 0.9 | 70 | | | | | | | | |
| 18 0.000 0.150 0.310 0.500 0.710 0.900 0.980 1.100 1.180 1.2 | 60 | | | | | | | | |
| 20 0.000 0.200 0.400 0.640 0.900 1.130 1.240 1.390 1.500 1.6 | 00 | | | | | | | | |
| 22 0.000 0.250 0.500 0.800 1.120 1.430 1.550 1.740 1.880 2.0 | 08 | | | | | | | | |

Tabel A. 8 Nilai koefisient $drag(C_D)$ pendekatan eksperimental untuk tiap tingkat perubahan sudut defleksi *rudder* pada 7 tingkat kecepatan

| U | | Nilai koefisien <i>drag</i> (C _D) stabiliser dan <i>rudder</i> | | | | | | | | |
|-------|-------|--|-------|-------|-------|-------|-------|-------|-------|-------|
| (m/s) | δ | | | | | | | | | |
| | 0 | 5 | 10 | 15 | 20 | 25 | 30 | 35 | 40 | 45 |
| 10 | 0,644 | 0,698 | 0,770 | 0,842 | 0,893 | 0,963 | 1,023 | 1,049 | 1,082 | 1,090 |
| 12 | 0,596 | 0,647 | 0,710 | 0,781 | 0,842 | 0,914 | 0,975 | 1,010 | 1,046 | 1,060 |
| 14 | 0,548 | 0,594 | 0,673 | 0,732 | 0,797 | 0,884 | 0,928 | 0,971 | 1,009 | 1,024 |
| 16 | 0,520 | 0,568 | 0,644 | 0,692 | 0,756 | 0,853 | 0,888 | 0,924 | 0,956 | 0,988 |
| 18 | 0,497 | 0,539 | 0,611 | 0,656 | 0,712 | 0,812 | 0,859 | 0,889 | 0,930 | 0,969 |
| 20 | 0,467 | 0,509 | 0,577 | 0,620 | 0,670 | 0,770 | 0,810 | 0,848 | 0,906 | 0,938 |
| 22 | 0,443 | 0,481 | 0,545 | 0,604 | 0,646 | 0,729 | 0,775 | 0,829 | 0,875 | 0,910 |
| | | | | | | | | | | |

| U | Nilai koefisien <i>side</i> (C _s) stabiliser dan <i>rudder</i> | | | | | | | | | |
|-------|--|-------|-------|-------|-------|-------|-------|-------|-------|-------|
| (m/s) | δ | | | | | | | | | |
| | 0 | 5 | 10 | 15 | 20 | 25 | 30 | 35 | 40 | 45 |
| 10 | 0,000 | 0,437 | 0,880 | 1,152 | 1,458 | 1,541 | 1,478 | 1,542 | 1,506 | 1,483 |
| 12 | 0,000 | 0,505 | 0,917 | 1,230 | 1,499 | 1,605 | 1,539 | 1,571 | 1,569 | 1,514 |
| 14 | 0,000 | 0,594 | 0,954 | 1,266 | 1,519 | 1,637 | 1,566 | 1,626 | 1,609 | 1,580 |
| 16 | 0,000 | 0,625 | 0,988 | 1,315 | 1,599 | 1,705 | 1,643 | 1,687 | 1,655 | 1,653 |
| 18 | 0,000 | 0,674 | 1,052 | 1,367 | 1,631 | 1,783 | 1,719 | 1,746 | 1,714 | 1,696 |
| 20 | 0,000 | 0,728 | 1,100 | 1,417 | 1,674 | 1,813 | 1,762 | 1,787 | 1,765 | 1,744 |
| 22 | 0,000 | 0,752 | 1,136 | 1,464 | 1,722 | 1,897 | 1,820 | 1,848 | 1,828 | 1,809 |

Tabel A. 9 Nilai koefisient $drag(C_D)$ pendekatan eksperimental untuk tiap tingkatperubahan sudut defleksi *rudder* pada 7 tingkat kecepatan

Tabel A. 10 Visualisasi aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 0^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s







Lanjutan tabel A. 10 Visualisasi aliran melintasi *stabilizer* dan *rudder* untuk sudut $(\delta = 0^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Tabel A. 11 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 5^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s**U (m/s)Profil Aliran Sesaat**



Lanjutan tabel A. 12 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk
sudut $(\delta = 5^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat







Tabel A. 12 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 10^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Lanjutan tabel A. 12 Karakteristik aliran melintasi stabilizer dan rudder untuk sudut $(\delta = 10^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Tabel A. 13 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut $(\delta = 15^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Tabel A. 14 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 20^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Lanjutan tabel A. 14 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk
sudut ($\delta = 20^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat







Tabel A. 15 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 25^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat







Lanjutan tabel A. 15 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk
sudut ($\delta = 25^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Tabel A. 16 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 30^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Tabel A. 16 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 30^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Tabel A. 17 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 35^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Lanjutan tabel A. 17 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk
sudut ($\delta = 35^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Tabel A. 18 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta =$ 40°) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s U (m/s) **Profil Aliran Sesaat**



Lanjutan tabel A. 18 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk
sudut ($\delta = 40^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/sU (m/s)Profil Aliran Sesaat



Tabel A. 19 Karakteristik aliran melintasi *stabilizer* dan *rudder* untuk sudut ($\delta = 45^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s









Tabel A. 20 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 0^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Lanjutan tabel A. 20 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 0^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 21 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 5^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 21 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 5^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 22 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 10^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 22 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 10^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Tabel A. 23 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 15^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Lanjutan tabel A. 23 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 15^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 24 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 20^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 24 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 20^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 25 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 25^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 25 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 25^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Tabel A. 26 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 30^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Lanjutan tabel A. 26 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 30^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 27 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 35^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 27 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 35^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Tabel A. 28 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 40^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s





Lanjutan tabel A. 28 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut $(\delta = 40^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Tabel A. 29 Profil kontur tekanan pada *stabilizer* dan *rudder* untuk sudut ($\delta = 45^{\circ}$) dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s



Contours of Static Pressure (pascal)

Oct 31, 2024 FLUENT 6.3 (3d, pbns, ske)

Lanjutan tabel A. 29 Profil kontur tekanan pada stabilizer dan rudder untuk sudut $(\delta = 45^{\circ})$ dengan kecepatan U = 12 m/s, 14 m/s, 18 m/s, dan 20 m/s

Lampiran B



Gambar B. 1 Model spesimen pwngujian wind tunnel



Gambar B. 2 Model spesimen pwngujian smoke flow



Gambar B. 3 Proses pengujian wind tunnel



Gambar B. 4 Proses pengujian smoke flow



Gambar B. 5 Proses pengujian CFD



Gambar B. 6 Proses presentasi artikel Conference; The 2 nd International Conference on Research in Engineering Science Technology (IC-REST) 2024